

## 21 Control Theory

This chapter more appropriately belongs in a book titled *The Nature of Mathematical Manipulation*, but it is here because the goal of modeling is rarely just the edification of the modeler: usually the intent is to act on the results of the model to achieve a desired goal. *Control theory* seeks to vary inputs to a system to produce desired outputs, a process that can range from years to femtoseconds [Rabitz *et al.*, 2000].

oven on-off proportional PID

state-space

Chapter 5 equations of motion Chapter 7 numerically solve

inverted pendulum

Problem 5.3

$$l\ddot{\theta} + (g + \ddot{z})\theta = 0 \quad . \quad (21.1)$$

$$\frac{d}{dt} \begin{bmatrix} \theta \\ \dot{\theta} \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ (g + \ddot{z})/l & 0 \end{bmatrix} \begin{bmatrix} \theta \\ \dot{\theta} \end{bmatrix} \quad (21.2)$$

parametric

discrete time

$$\begin{aligned} \vec{x}_{n+1} &= \mathbf{A}\vec{x}_n + \mathbf{B}\vec{u}_n \\ \vec{y}_n &= \mathbf{C}\vec{x}_n + \mathbf{D}\vec{u}_n \end{aligned} \quad (21.3)$$

solve

$$\begin{aligned} \vec{x}_1 &= \mathbf{A}\vec{x}_0 + \mathbf{B}\vec{u}_0 \\ \vec{x}_2 &= \mathbf{A}\vec{x}_1 + \mathbf{B}\vec{u}_1 \\ &= \mathbf{A}[\mathbf{A}\vec{x}_0 + \mathbf{B}\vec{u}_0] + \mathbf{B}\vec{u}_1 \\ &= \mathbf{A}^2\vec{x}_0 + \mathbf{A}\mathbf{B}\vec{u}_0 + \mathbf{B}\vec{u}_1 \\ &\vdots \\ \vec{x}_n &= \mathbf{A}^n\vec{x}_0 + \mathbf{B}\vec{u}_{n-1} + \mathbf{A}\mathbf{B}\vec{u}_{n-2} + \dots + \mathbf{A}^{n-1}\mathbf{B}\vec{u}_0 \end{aligned} \quad (21.4)$$

stability eigenvalues magnitude  $>$ ,  $<$  1

continuous time

$$\begin{aligned}\dot{\vec{x}} &= \mathbf{A}\vec{x} + \mathbf{B}\vec{u} \\ \vec{y} &= \mathbf{C}\vec{x} + \mathbf{D}\vec{u}\end{aligned}\tag{21.5}$$

matrix exponential

$$e^{\mathbf{A}t} = \mathbf{I} + \mathbf{A}t + \frac{1}{2!}\mathbf{A}^2t^2 + \dots\tag{21.6}$$

solution

$$\vec{x}(t) = e^{\mathbf{A}t}\vec{x}_0 + \int_0^t e^{\mathbf{A}(t-\tau)}\mathbf{B}\vec{u}(\tau) d\tau\tag{21.7}$$

stability eigenvalues real part  $>$ ,  $<$  0  
Laplace transform

$$\begin{aligned}s\vec{X}(s) &= \mathbf{A}\vec{X}(s) + \mathbf{B}\vec{U}(s) \\ \vec{X}(s) &= (s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B}\vec{U}(s)\end{aligned}\tag{21.8}$$

$$\vec{Y}(s) = (\mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B} + \mathbf{D})\vec{U}(s)\tag{21.9}$$

$$\vec{Y}(s) = \mathbf{G}(s)\vec{U}(s)$$

$$\mathbf{G}(s) = \mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B} + \mathbf{D}\tag{21.10}$$

transfer function poles = system eigenvalues  
controllability  
discrete time

$$\vec{U}_{n-1} \equiv \begin{bmatrix} \vec{u}_{n-1} \\ \vdots \\ \vec{u}_0 \end{bmatrix}\tag{21.11}$$

$$\mathbf{P}_{n-1} \equiv [\mathbf{B} \ \mathbf{A}\mathbf{B} \ \dots \ \mathbf{A}^{n-1}\mathbf{B}]\tag{21.12}$$

$$\vec{x}_n = \mathbf{A}^n\vec{x}_0 + \mathbf{P}_{n-1}\vec{U}_{n-1}\tag{21.13}$$

$\mathbf{P}$  full row rank of  $\vec{x} = N$   
saturates at  $n = N$  from Cayley-Hamilton (every matrix satisfies its own characteristic equation)  
divide by  $\Delta(\mathbf{A})$ :

$$P(\mathbf{A}) = Q(\mathbf{A})\Delta(\mathbf{A}) + R(\mathbf{A})\tag{21.14}$$

continuous time:

$$\begin{aligned}
\vec{x}(t) - e^{At}\vec{x}_0 &= \int_0^t e^{A(t-\tau)}\mathbf{B}\vec{u}(\tau) d\tau \\
&= \int_0^t \left[ \mathbf{I} + \mathbf{A}(t-\tau) + \dots + \frac{1}{(N-1)!}\mathbf{A}^{N-1}(t-\tau)^{N-1} \right] \mathbf{B}\vec{u}(\tau) d\tau \\
&= \sum_{n=0}^{N-1} \mathbf{A}^n \mathbf{B} \int_0^t \frac{1}{n!} (t-\tau)^n \vec{u}(\tau) d\tau \\
&\equiv \sum_{n=0}^{N-1} \mathbf{A}^n \mathbf{B} \vec{v}_n \\
&= [\mathbf{B} \quad \mathbf{A}\mathbf{B} \quad \dots \quad \mathbf{A}^{N-2}\mathbf{B} \quad \mathbf{A}^{N-1}\mathbf{B}] \begin{bmatrix} \vec{v}_0 \\ \vec{v}_1 \\ \vdots \\ \vec{v}_{N-1} \end{bmatrix} \tag{21.15}
\end{aligned}$$

observability

$$\vec{y}_n = \mathbf{C}\mathbf{A}^n \vec{x}_0 + \mathbf{C}\mathbf{B}\vec{u}_{n-1} + \mathbf{C}\mathbf{A}\mathbf{B}\vec{u}_{n-2} + \dots + \mathbf{C}\mathbf{A}^{n-1}\mathbf{B}\vec{u}_0 + \mathbf{D}\vec{u}_n \tag{21.16}$$

$$\mathbf{C}\mathbf{A}^n \vec{x}_0 = \vec{y}_n - \mathbf{C}\mathbf{B}\vec{u}_{n-1} + \mathbf{C}\mathbf{A}\mathbf{B}\vec{u}_{n-2} + \dots + \mathbf{C}\mathbf{A}^{n-1}\mathbf{B}\vec{u}_0 + \mathbf{D}\vec{u}_n \tag{21.17}$$

know right hand side

$$\mathbf{Q}_{a-1} \equiv \begin{bmatrix} \mathbf{C} \\ \mathbf{C}\mathbf{A} \\ \mathbf{C}\mathbf{A}^2 \\ \vdots \\ \mathbf{C}\mathbf{A}^{n-1} \end{bmatrix} \tag{21.18}$$

full column rank  $N$

$\vec{u} = \mathbf{0}$ :

$$\begin{aligned}
\vec{y}(t) &= \mathbf{C}e^{At}\vec{x}_0 \\
&= \mathbf{C} \left[ \mathbf{I} + \dots + \frac{1}{(N-1)!}\mathbf{A}^{N-1}t^{N-1} \right] \vec{x}_0 \\
&= \sum_{n=0}^{N-1} \mathbf{C}\mathbf{A}^n \frac{1}{n!} t^n \vec{x}_0 \\
&= [\mathbf{C} \quad \mathbf{C}\mathbf{A} \quad \dots \quad \mathbf{C}\mathbf{A}^{N-2} \quad \mathbf{C}\mathbf{A}^{N-1}] \begin{bmatrix} \vec{x}_0 \\ t\vec{x}_0 \\ \vdots \\ \frac{1}{(N-1)!}t^{N-1}\vec{x}_0 \end{bmatrix} \tag{21.19}
\end{aligned}$$

output feedback  
state variable feedback  
feedback gain matrix  $\mathbf{K}$

$$\vec{u} = \vec{v} - \mathbf{K}\vec{x} \quad (21.20)$$

$$\dot{\vec{x}} = (\mathbf{A} - \mathbf{BK})\vec{x} + \mathbf{B}\vec{v} \quad (21.21)$$

$$|\lambda\mathbf{I} - \mathbf{A} + \mathbf{BK}| = 0 \quad (21.22)$$

expand, equate to desired characteristic polynomial, match orders  $\lambda$   
pole-zero placement, cancellation  
more complex for multiple inputs

$$(\mathbf{A} - \mathbf{BK})\vec{\varphi}_i = \lambda_i\vec{\varphi}_i \quad (21.23)$$

$$[(\lambda_i\mathbf{I} - \mathbf{A}) \mathbf{B}] \begin{bmatrix} \vec{\varphi}_i \\ \mathbf{K}\vec{\varphi}_i \end{bmatrix} = \vec{0} \quad (21.24)$$

pick  $\{\lambda_i\}$ , find basis null space, solve for  $\mathbf{K}$   
Lyapunov  
stability  
 $v = \vec{x}^T \mathbf{P} \vec{x}, \mathbf{P} < 0$

$$\begin{aligned} \dot{v} &= \dot{\vec{x}}^T \mathbf{P} \vec{x} + \vec{x}^T \mathbf{P} \dot{\vec{x}} \\ &= \vec{x}^T \mathbf{A}^T \mathbf{P} \vec{x} + \vec{x}^T \mathbf{P} \mathbf{A} \vec{x} \\ &= \vec{x}^T (\mathbf{A}^T \mathbf{P} + \mathbf{P} \mathbf{A}) \vec{x} \end{aligned} \quad (21.25)$$

$v$  decreasing:  
 $(\mathbf{A}^T \mathbf{P} + \mathbf{P} \mathbf{A}) < 0$   
LMI, feasibility  
 $\dot{\vec{x}} = \mathbf{A}\vec{x}$   
 $\vec{u} = \mathbf{K}\vec{x}$

$$(\mathbf{A} + \mathbf{BK})^T \mathbf{P} + \mathbf{P}(\mathbf{A} + \mathbf{BK}) < 0 \quad (21.26)$$

convert to LMI

$$\mathbf{Q}(\mathbf{A} + \mathbf{BK})^T + (\mathbf{A} + \mathbf{BK})\mathbf{Q} < 0 \quad (21.27)$$

substitute  $\mathbf{K} = \mathbf{YQ}^{-1}$

$$\mathbf{AQ} + \mathbf{QA}^T + \mathbf{BY} + \mathbf{Y}^T \mathbf{B}^T < 0 \quad (21.28)$$

jointly find controller, stability

optimal: limit, optimize resources  
 adaptive: online modeling  
 robust: insensitive to model assumptions, system changes, faulty operation  
 nonlinear: geometrical, convex  
 Schur complement

$$\begin{pmatrix} \mathbf{Q} & \mathbf{S} \\ \mathbf{S}^T & \mathbf{R} \end{pmatrix} > 0 \quad (21.29)$$

$$\mathbf{R} > 0, \mathbf{Q} - \mathbf{S}\mathbf{R}^{-1}\mathbf{S}^T > 0$$

Refs

Brogan: Modern Control Theory

Bertsekas: Nonlinear Programming

Zhou, Doyle, Glover: Robust and Optimal Control

Boyd, El Ghaoui, Feron, Balakrishnan: Linear Matrix Inequalities in System and Control Theory

Boyd and Vandenberghe: Convex Optimization

LQR

binary

$$\min |y(t) - y_r(t)|^2$$

quadratic

$$\begin{bmatrix} \vec{u} \\ 1 \end{bmatrix}^T \begin{bmatrix} \mathbf{M} & \vec{v} \\ \vec{v}^T & s \end{bmatrix} \begin{bmatrix} \vec{u} \\ 1 \end{bmatrix} \quad (21.30)$$

$$u_i \in \{-1, +1\}$$

primal

combinatorial optimization

$$\min \vec{x}^T \mathbf{M} \vec{x}, x_i^2 - 1 = 0$$

Lagrangian

$$\begin{aligned} \mathcal{L}(\vec{x}, \lambda) &= \vec{x}^T \mathbf{M} \vec{x} - \sum_i \lambda_i (x_i^2 - 1) \\ &= \vec{x}^T (\mathbf{M} - \mathbf{\Lambda}) \vec{x} + \text{Tr} \mathbf{\Lambda} \end{aligned} \quad (21.31)$$

dual

$$\max \text{Tr} \mathbf{\Lambda}, \mathbf{M} - \mathbf{\Lambda} \succeq 0$$

continuous

Riccati

small gain theorem Kalman-Yakubovich-Popov Lemma H2 Hinf mu structured singular value

nonlinear

SDP Schur complement global linearization, time-varying linear Lie bracket quantum

compute, control

distributed

subgradients, dual

LMI, feasibility  
min norm control:

$$\vec{U}_n = -\mathbf{P}_n^T [\mathbf{P}_n \mathbf{P}_n^T]^{-1} \mathbf{A}^n \vec{x}_0 \quad (21.32)$$

$$\begin{bmatrix} \mathbf{C}^T & \mathbf{A}^T \mathbf{C}^T & \cdots & \mathbf{A}^{N-2T} \mathbf{C}^T & \mathbf{A}^{N-1T} \mathbf{C}^T \end{bmatrix} \quad (21.33)$$

State variable feedback  
discrete time

$$\vec{u}_n = \mathbf{F} \vec{v}_n - \mathbf{K} \vec{x}_n \quad (21.34)$$

$$\vec{x}_{n+1} = [\mathbf{A} - \mathbf{BK}] \vec{x}_n + \mathbf{B} \mathbf{F} \vec{v}_n \quad (21.35)$$

$$\Delta(\lambda) = |\lambda \mathbf{I} - \mathbf{A} + \mathbf{BK}| = 0 \quad (21.36)$$

$$[\mathbf{A} - \mathbf{BK}] \vec{v}_i = \lambda_i \vec{v}_i \quad (21.37)$$

$$[\lambda_i \mathbf{I} - \mathbf{A} \quad \mathbf{B}] \begin{bmatrix} \vec{v}_i \\ \mathbf{K} \vec{v}_i \end{bmatrix} = \mathbf{0} \quad (21.38)$$

SVD basis nullspace  
SVD solve for  $\mathbf{K}$   
continuous time

$$\begin{aligned} \dot{\vec{x}} &= \mathbf{A} \vec{x} + \mathbf{B}(\vec{u} - \mathbf{K} \vec{x}) \\ &= (\mathbf{A} - \mathbf{BK}) \vec{x} + \mathbf{B} \vec{u} \end{aligned} \quad (21.39)$$

$$\begin{aligned} |\lambda \mathbf{I} - \mathbf{A} + \mathbf{BK}| &= |(\lambda \mathbf{I} - \mathbf{A})[\mathbf{I} + (\lambda \mathbf{I} - \mathbf{A})^{-1} \mathbf{BK}]| \\ &= |\lambda \mathbf{I} - \mathbf{A}| |\mathbf{I} + (\lambda \mathbf{I} - \mathbf{A})^{-1} \mathbf{BK}| \end{aligned} \quad (21.40)$$

controllability implies sufficient rank  
SVD  
Observers

$$\vec{x}'_{n+1} = \mathbf{A}' \vec{x}'_n + \mathbf{E} \vec{y}_n + \vec{d}_n \quad (21.41)$$

$$\vec{e}_n = \vec{x}_n - \vec{x}'_n \quad (21.42)$$

$$\vec{e}_{n+1} = \mathbf{A}\vec{x}_n + \mathbf{B}\vec{u}_n - \mathbf{A}'\vec{x}'_n - \mathbf{E}\mathbf{C}\vec{x}_n - \mathbf{E}\mathbf{D}\vec{u}_n - \vec{d}_n \quad (21.43)$$

$$\vec{d} = (\mathbf{B} - \mathbf{E}\mathbf{D})\vec{u}_n \quad (21.44)$$

$$\mathbf{A}' = \mathbf{A} - \mathbf{E}\mathbf{C} \quad (21.45)$$

$$\vec{e}_{n+1} = \mathbf{A}'\vec{e}_n \quad (21.46)$$

$$|\lambda\mathbf{I} - \mathbf{A} + \mathbf{E}\mathbf{C}| = 0 \quad (21.47)$$

system identification impulse frequency correlation  
 state estimation observer  
 optimal  
 dynamic programming  
 vector field  
 tangent space  
 Lie bracket  
 algebra  
 linear  
 car  
 linear car  
 car bracket  
 feedback linearization eliminate higher-order terms  
 non-holonomic (velocity constraint) Nyquist stability Bode plot lag, lead compensation  
 ref to adaptive steppers Desoer's famous 1969 paper on the stability of slowly-varying  
 systems controllability wronskian

## 21.1 PROBLEMS

(21.1) inverted pendulum

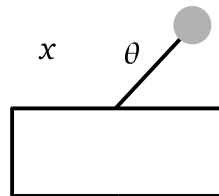


Figure 21.1. Cart

(21.2) PID problem

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NMM Programming  
Languages  
symbolic mathematica maple interpreted APL Python scipy byte-code Java dyadic  
compiled Fortran C assembly FPGA GPU HDL VHDL Verilog ASIC DEShaw ALA  
Libraries  
C libs direct indirect NR Netlib linalg blas numpy  
Graphics  
high-low  
python ipython matplotlib tk wx VTK scipy  
java  
gl  
x  
postscript  
\*ification  
sdl stl